

# AGENCY REPORTS

# JAXA

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JAXA/EORC

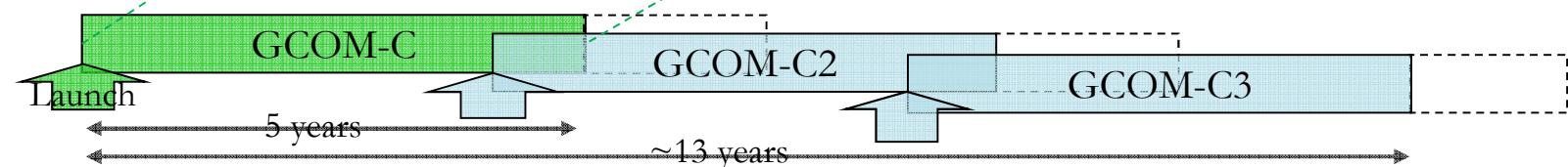
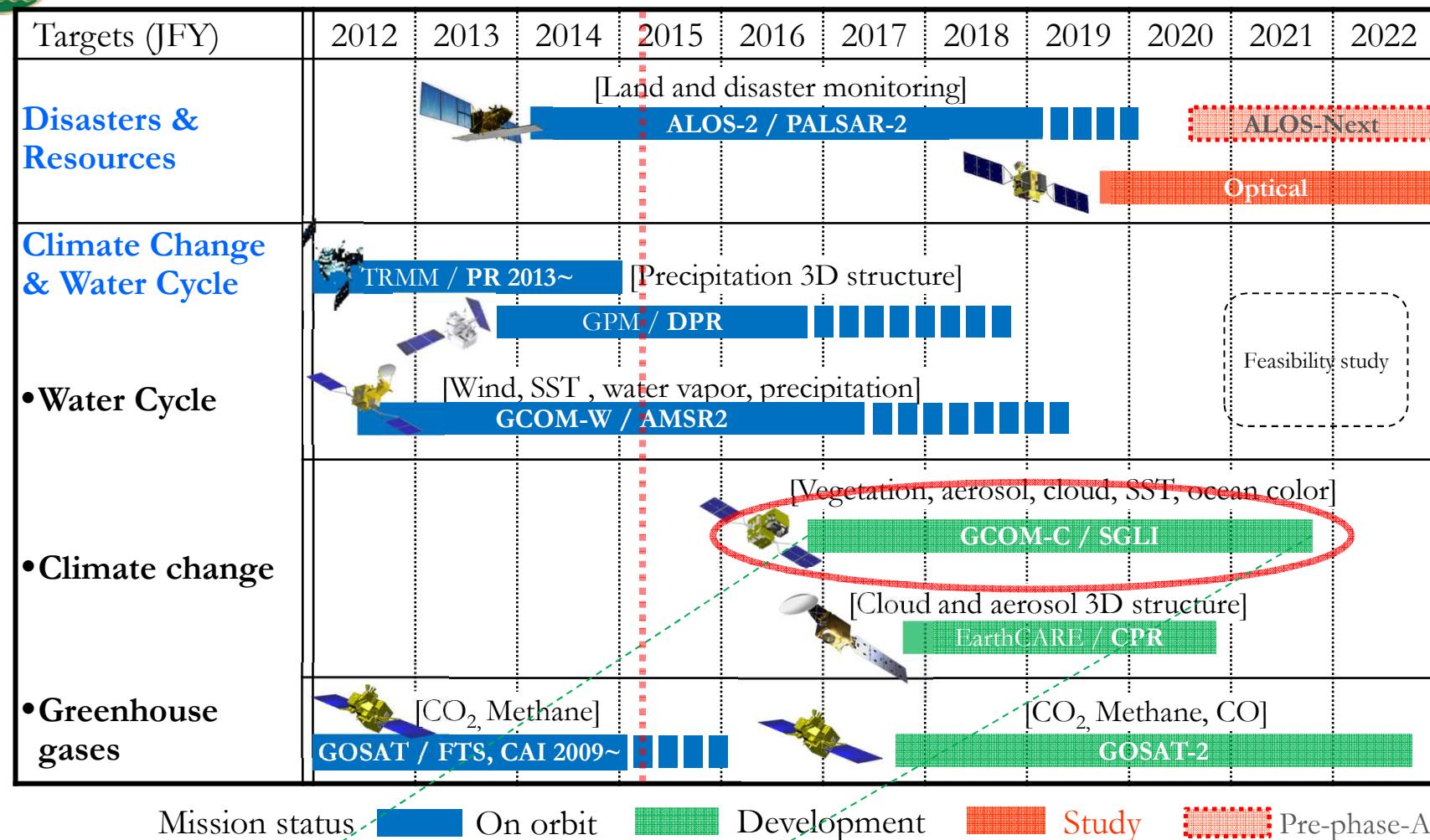


International Ocean Colour Science  
Meeting 2015

Advancing Global  
Ocean Colour  
Observations

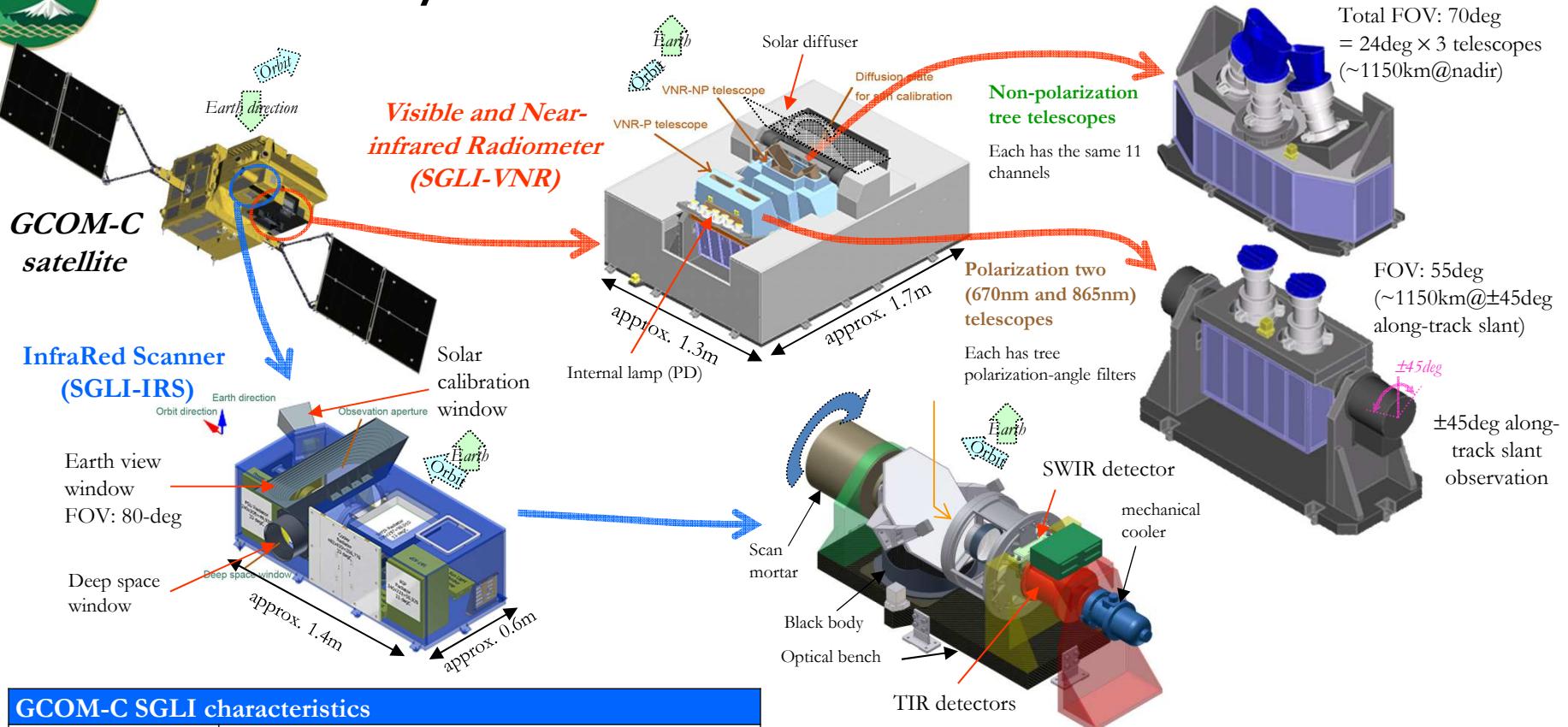


# JAXA Earth Observation Satellite missions





# GCOM-C/SGLI



GCOM-C SG LI characteristics	
Orbit	Sun-synchronous (descending local time: <b>10:30</b> ), Altitude: 798km, Inclination: 98.6deg
Launch Date	<b>JFY 2016</b>
Mission Life	5 years (3 satellites; total 13 years)
Scan	Push-broom electric scan (VNR: VN & P) Wisk-broom mechanical scan (IRS: SW & T)
Scan width	<b>1150km</b> cross track (VNR: VN & P) <b>1400km</b> cross track (IRS: SW & T)
Spatial resolution	<b>250m, 500m, 1km</b>
Polarization	<b>3 polarization angles for POL</b>
Along track tilt	Nadir for VN, SW and TIR, & +/-45 deg for P



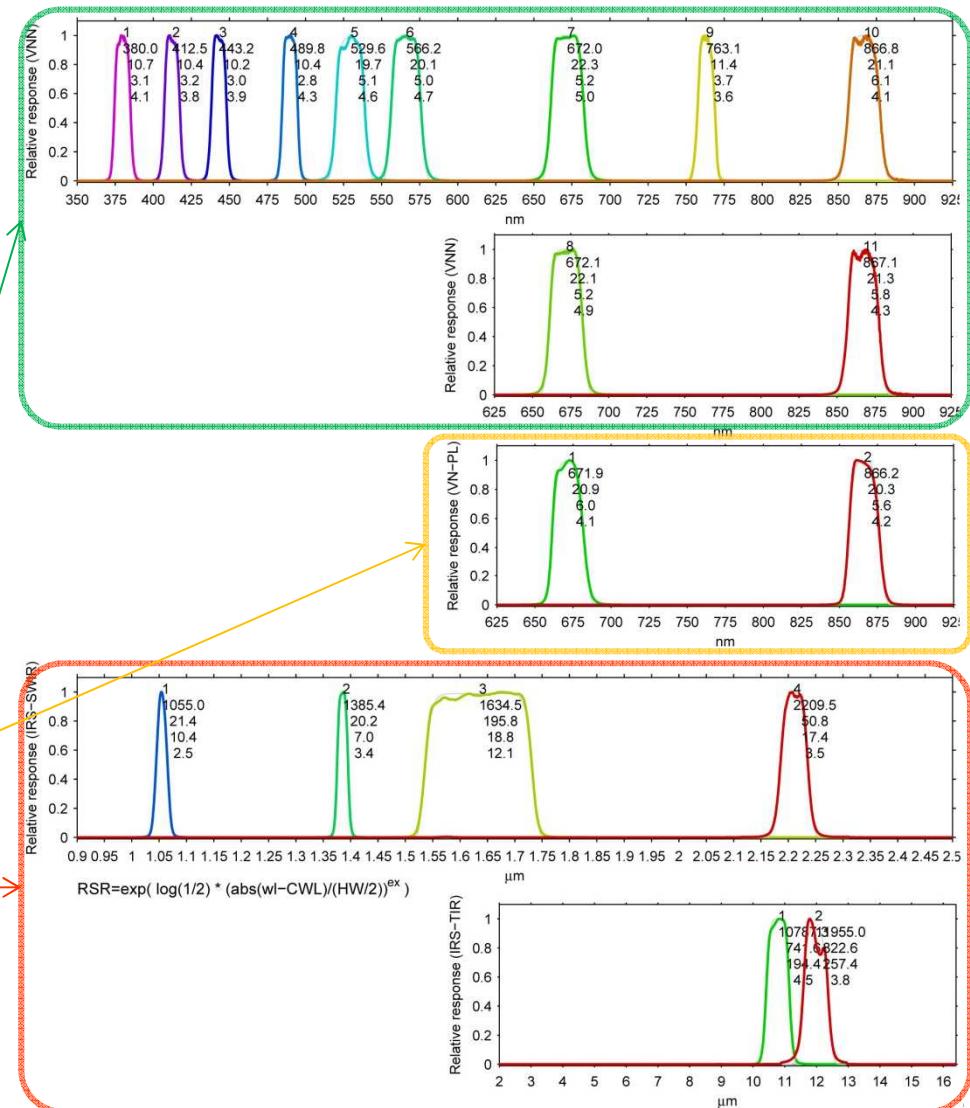


# SGLI Bands

Multi-angle obs. for 674nm and 869nm

Characteristics of SGLI spectral bands						
CH	$\lambda$	$\Delta\lambda$	$L_{std}$	$L_{max}$	SNR@ $L_{std}$	IFOV
	nm		W/m <sup>2</sup> /sr/ $\mu$ m K: Kelvin		K: NEΔT	m
VN1	380	10	60	210	250	<b>250 /1000</b>
VN2	412	10	75	250	400	<b>250 /1000</b>
VN3	443	10	64	400	300	<b>250 /1000</b>
VN4	490	10	53	120	400	<b>250 /1000</b>
VN5	530	20	41	350	250	<b>250 /1000</b>
VN6	565	20	33	90	400	<b>250 /1000</b>
VN7	673.5	20	23	62	400	<b>250 /1000</b>
VN8	673.5	20	25	210	250	<b>250 /1000</b>
VN9	763	12	40	350	1200*	<b>250 /1000*</b>
VN10	868.5	20	8	30	400	<b>250 /1000</b>
VN11	868.5	20	30	300	200	<b>250 /1000</b>
POL1	673.5	20	25	250	250	<b>1000</b>
POL2	868.5	20	30	300	250	<b>1000</b>
SW1	1050	20	57	248	500	<b>1000</b>
SW2	1380	20	8	103	150	<b>1000</b>
SW3	1630	200	3	50	57	<b>250 /1000</b>
SW4	2210	50	1.9	20	211	<b>1000</b>
TIR1	10800	700	300K	340K	0.2K	<b>250/500/1000</b>
TIR2	12000	700	300K	340K	0.2K	<b>250/500/1000</b>

[http://suzaku.eorc.jaxa.jp/GCOM\\_C/about/c\\_sgli\\_prod\\_01.html](http://suzaku.eorc.jaxa.jp/GCOM_C/about/c_sgli_prod_01.html)



Uchikawa, T., K. Tanaka, Y. Okamura, S. Tsuda, and T. Amano, "Proto Flight Model (PFM) performance and development status of Cisible and Near Infrared Radiometer (VNR) on the Second-generation Global Imager (SGLI)", SPIE Asia-Pacific Remote sensing, Beijing, China, 9264-27, 2014.

Tanaka, K., Y. Okamura, T. Amano, T. Hosokawa, and T. Uchikita, "The development status of Second Generation Global Imager Infrared Scanning Radiometer (SGLI-IRS)", SPIE Asia-Pacific Remote sensing, Beijing, China, 9264-15, October, 2014.

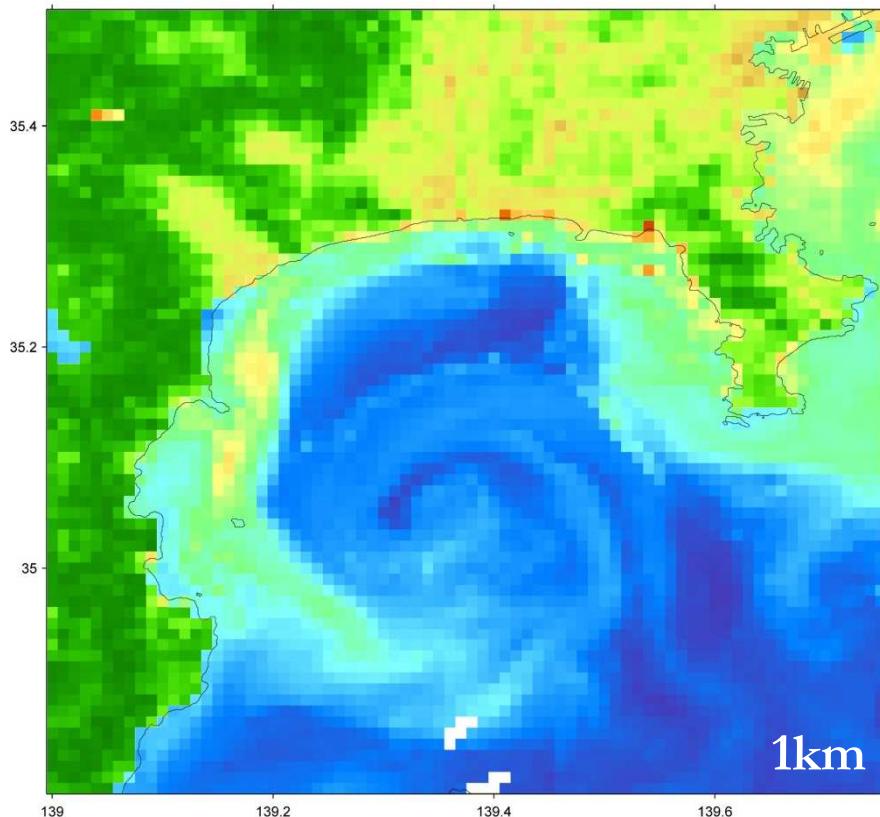


# SGLI 250m Visible-SWIR observation

250m resolution to detect finer structure in the coastal area such as river outflow, regional blooms, small current, and redtide

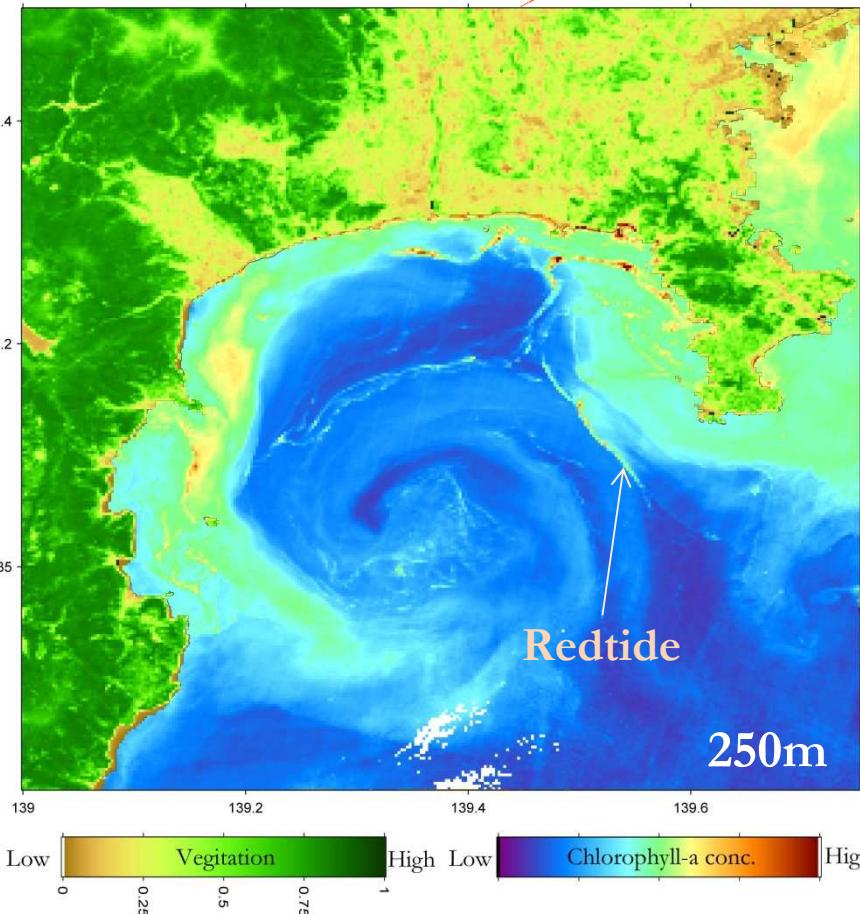


1-km resolution



(a) 250m Sagami-Bay (2 May 2015, CHL by LCI)

250-m resolution



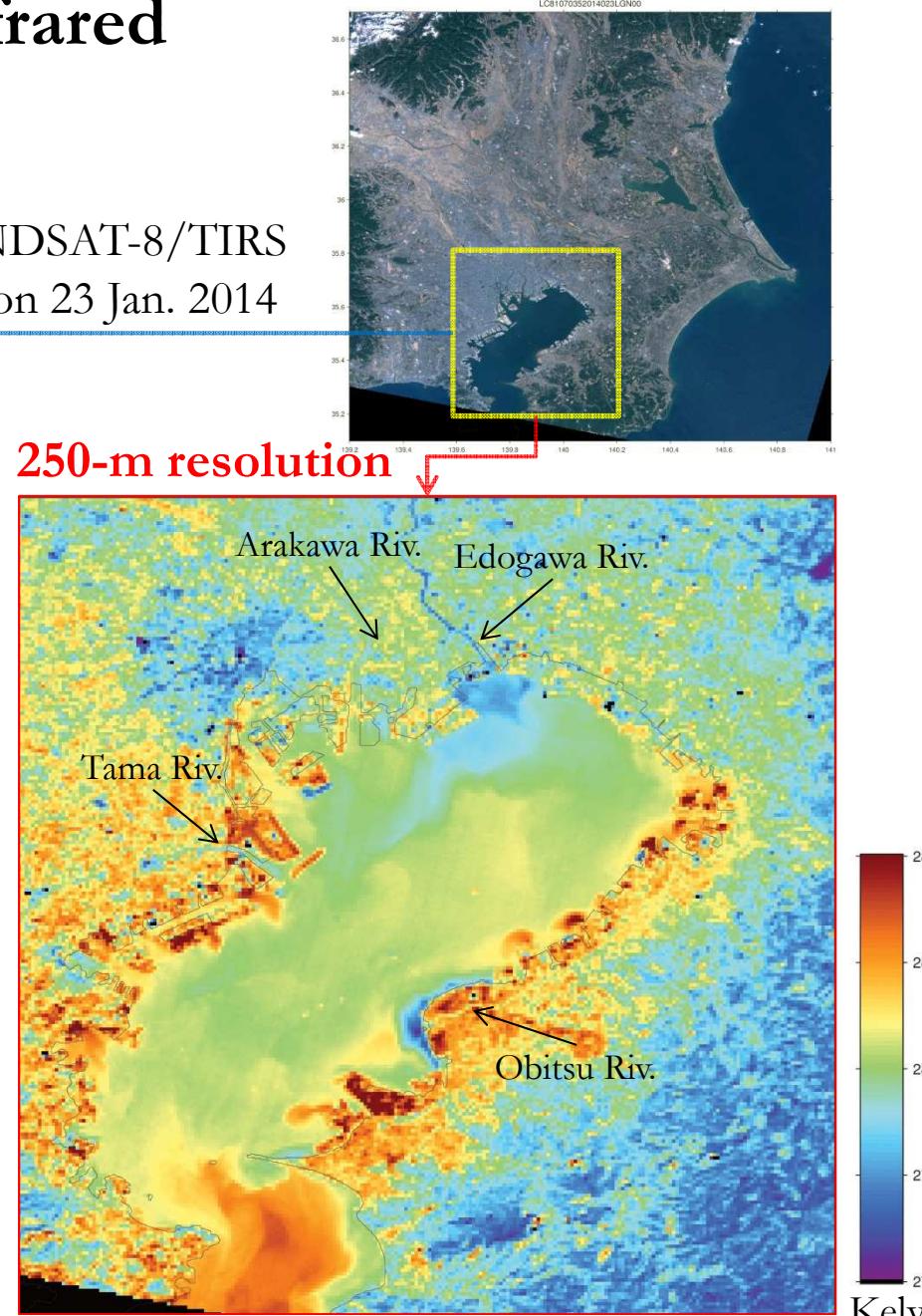
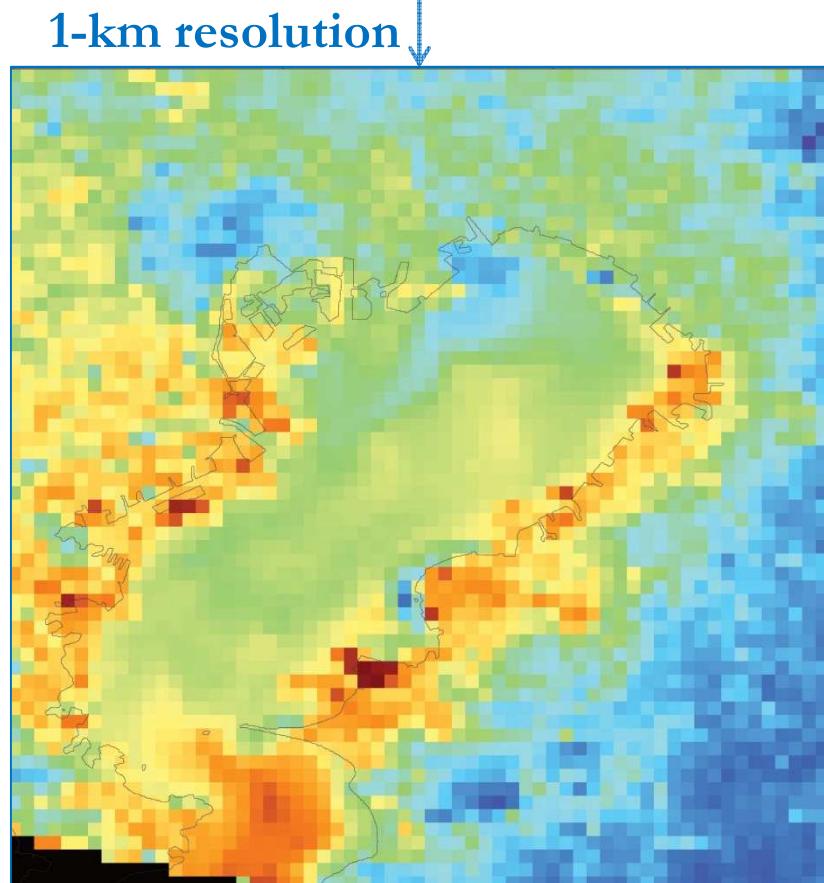
(b) 1km Sagami-Bay (2 May 2015, CHL by LCI)

250m SGLI products simulated by Landsat image on 2 May 2015



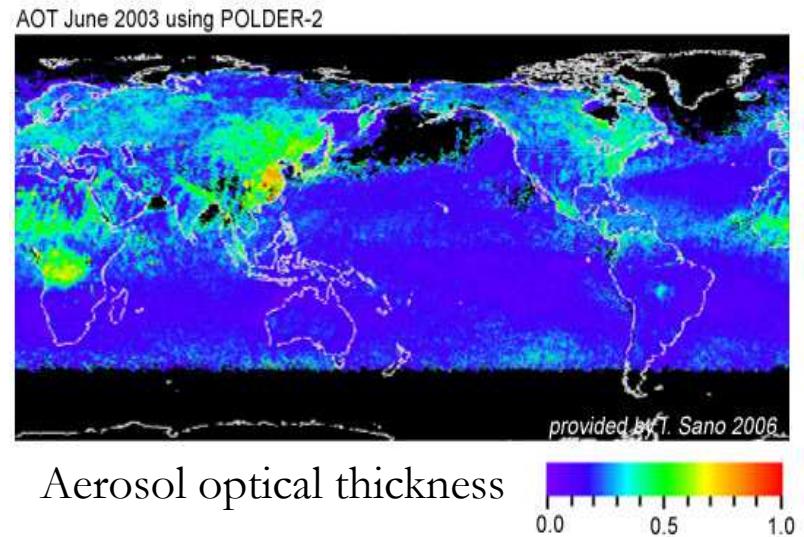
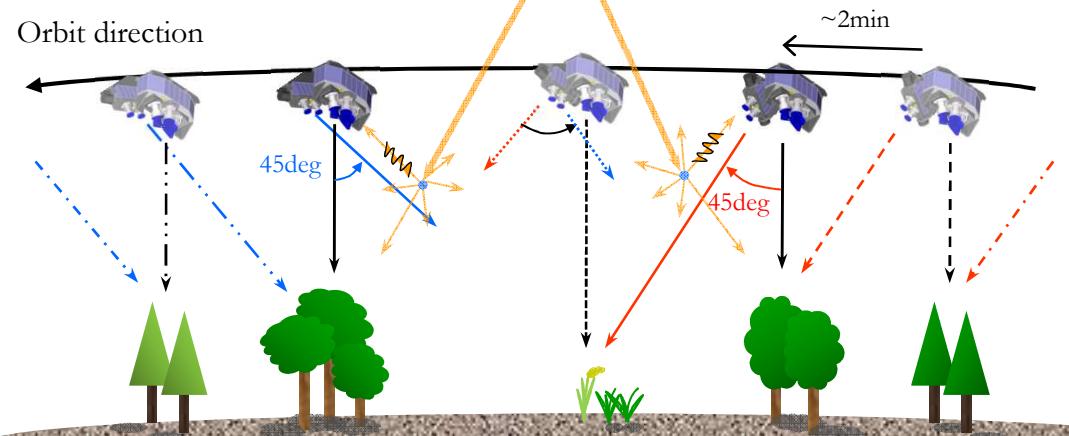
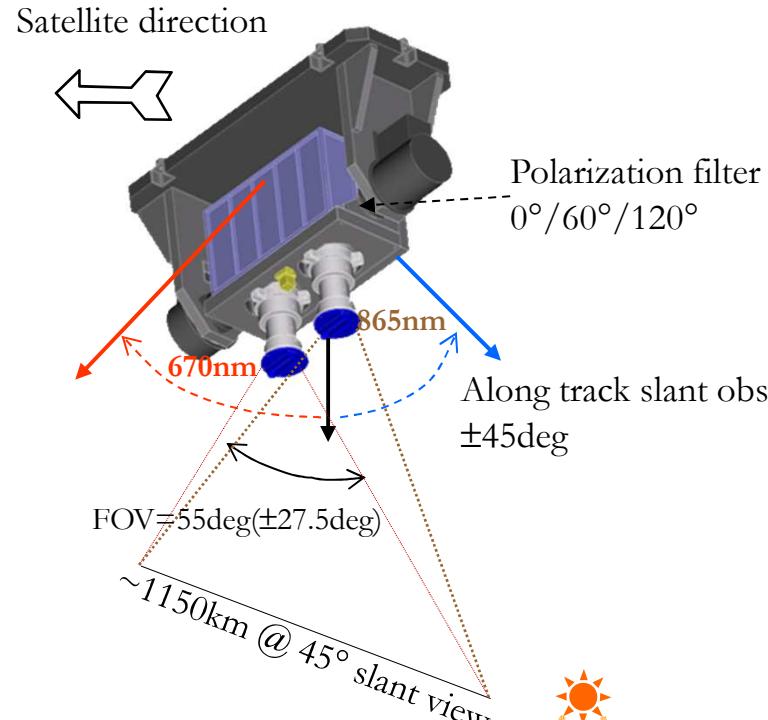
# SGLI 250m Thermal infrared observations

Simulated by LANDSAT-8/TIRS  
11 $\mu$ m 100m data on 23 Jan. 2014





# SGLI slant-view polarization observation



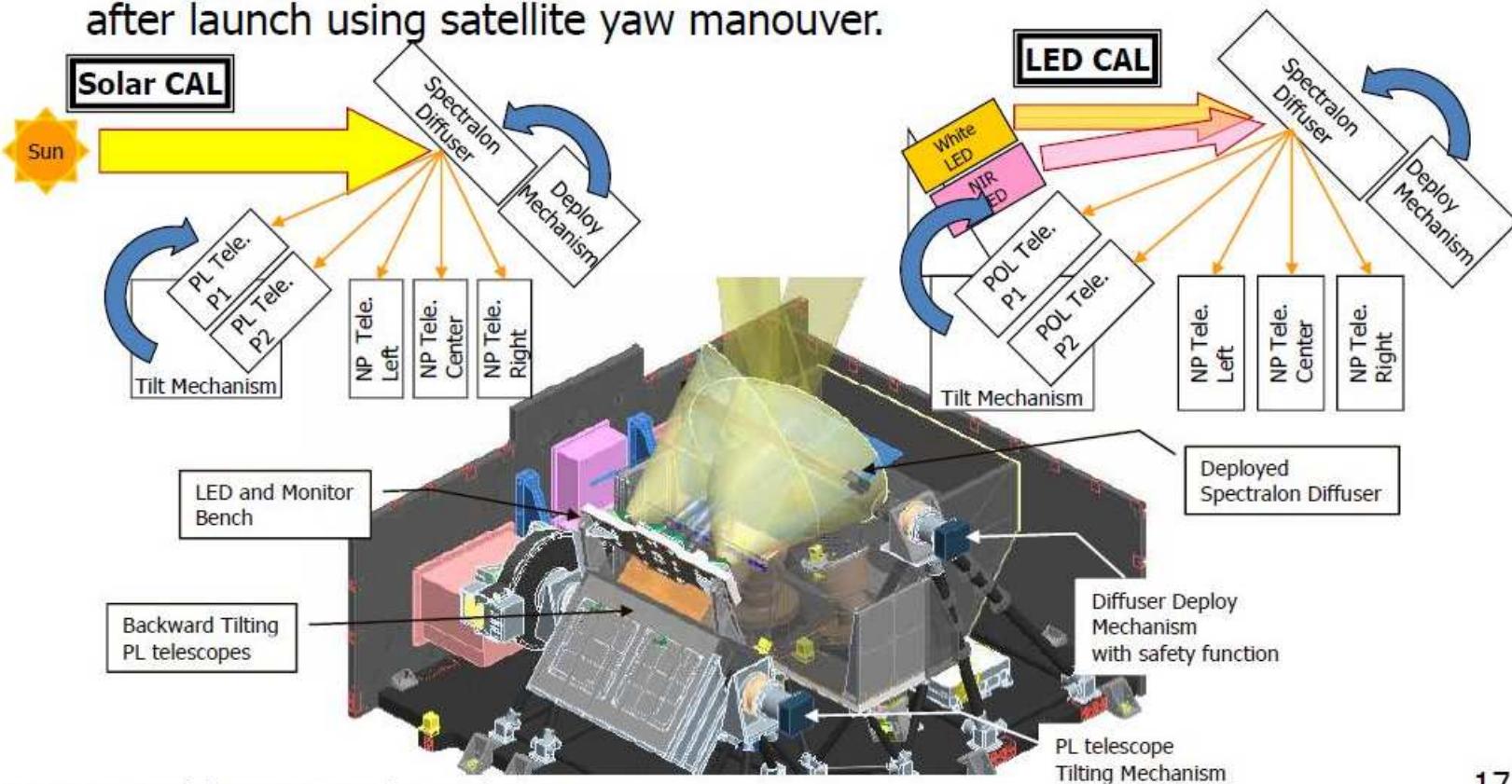
Global aerosol optical thickness in June 2003 using POLDER-2 polarization reflectance (provided by T. Sano, Kinki Univ.)



# VNR Onboard CALIBRATION



- Deployable Spectralon diffuser is used for both Solar and LED calibration. Calibration coefficient will be traced using this solar calibration data.
- $\beta$  angle dependency for solar calibration will be characterized shortly after launch using satellite yaw manouver.



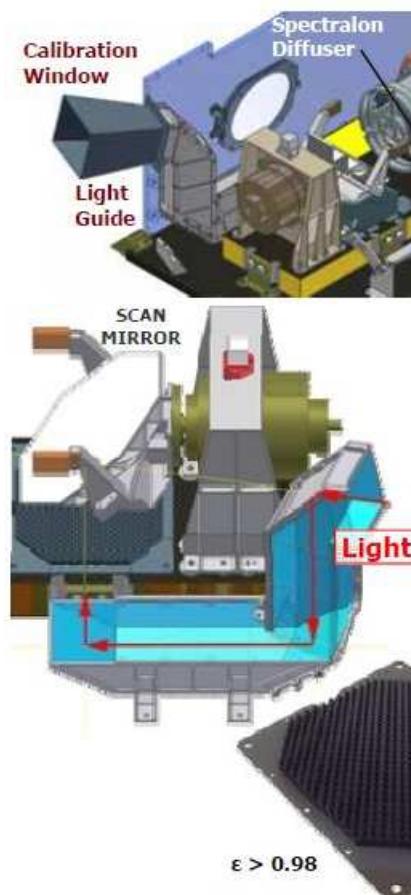
Operation Concept f the SGLI, SPIE Incheon, October 13, 2010



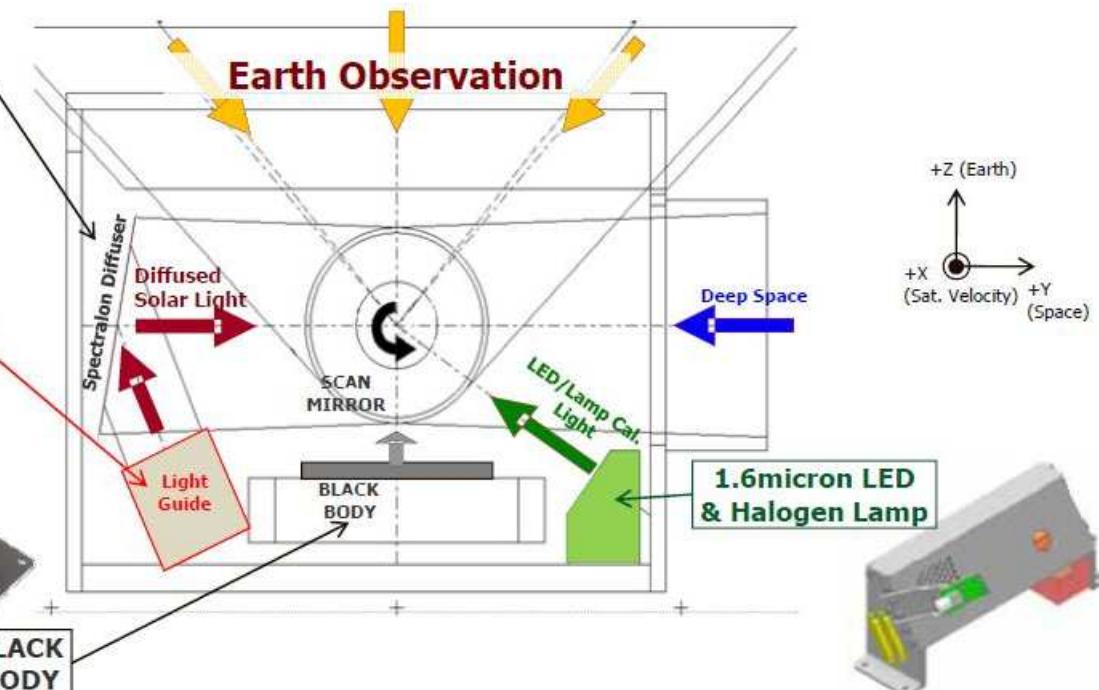
# IRS Onboard calibration



■ IRS 81rpm rotating for both "Earth Observation" and "Calibration".



TIR Calibration : "BLACK BODY" and "Deep Space"  
SWI Calibration : "Diffused Solar Light", "LED/Lamp" and "Deep Space"



Operation Concept of the SGLI, SPIE Incheon, October 13, 2010



# Calibration Manouver



## Lunar Calibration Manouver

- ✓ Moon reflecting solar light is a stable light source as a long term calibration reference of the optical sensors.
- ✓ Lunar calibration manouver is done at every 29 days interval (= synodic period of the moon and the sun)
- ✓ SGLI observes same phase angle moon (7deg) during 5 years mission for the integral lunar calibration.



ADEOS-II/GLI 250m

## Sun Angle Correction Manouver

- ✓ As earth rotates around the sun, solar direction gradually change because of the elliptical and inclined orbit of the earth.
- ✓ The local sun time tolerance is another factor for this solar angle change.
- ✓ Solar angle calibration manouver is done to establish the solar angle dependency database for this calibration.

## Obs. Angle Correction Manouver

- ✓ Because VNR uses wide FOV's pushbroom type CCD sensor, the satellite observation angle dependency is another key calibration factor.
- ✓ The satellite yaw axis is 90deg rotated to observe the almost same earth target with different CCD detectors.
- ✓ The homogeneous target is used for this calibration.

*Operation Concept f the SGLI, SPIE Incheon, October 13, 2010*

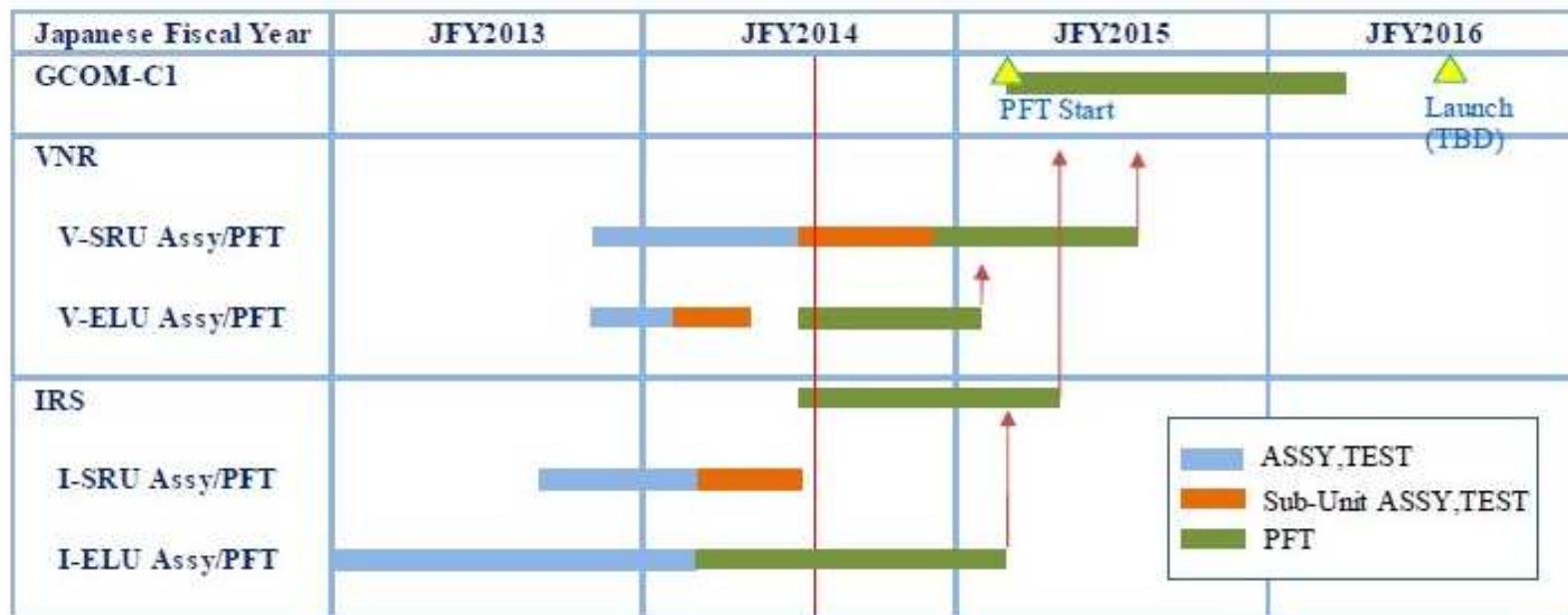
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[SGLI Operation Concept \(SPIE, Incheon, Oct. 13, 2010\)](#)



# GCOM-C/SGLI

- ✓ VNR (telescopes) and IRS (scanner) assembly and preflight tests have been done
- ✓ Satellite-system integration test will be started in this summer.

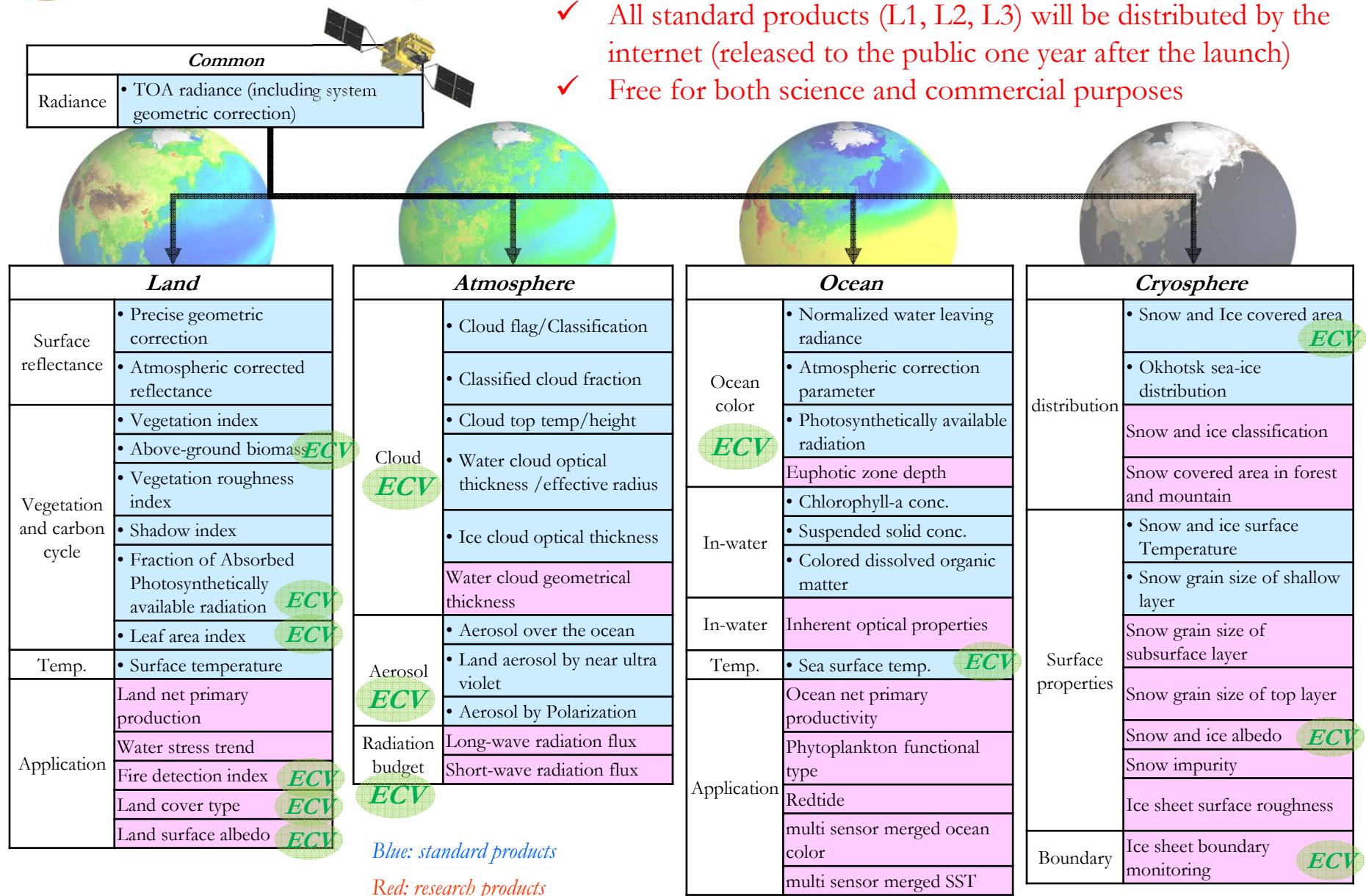


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### 3. GCOM-C Observation Product Development

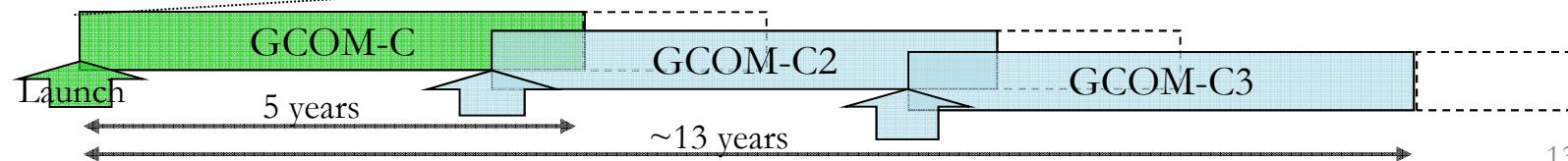
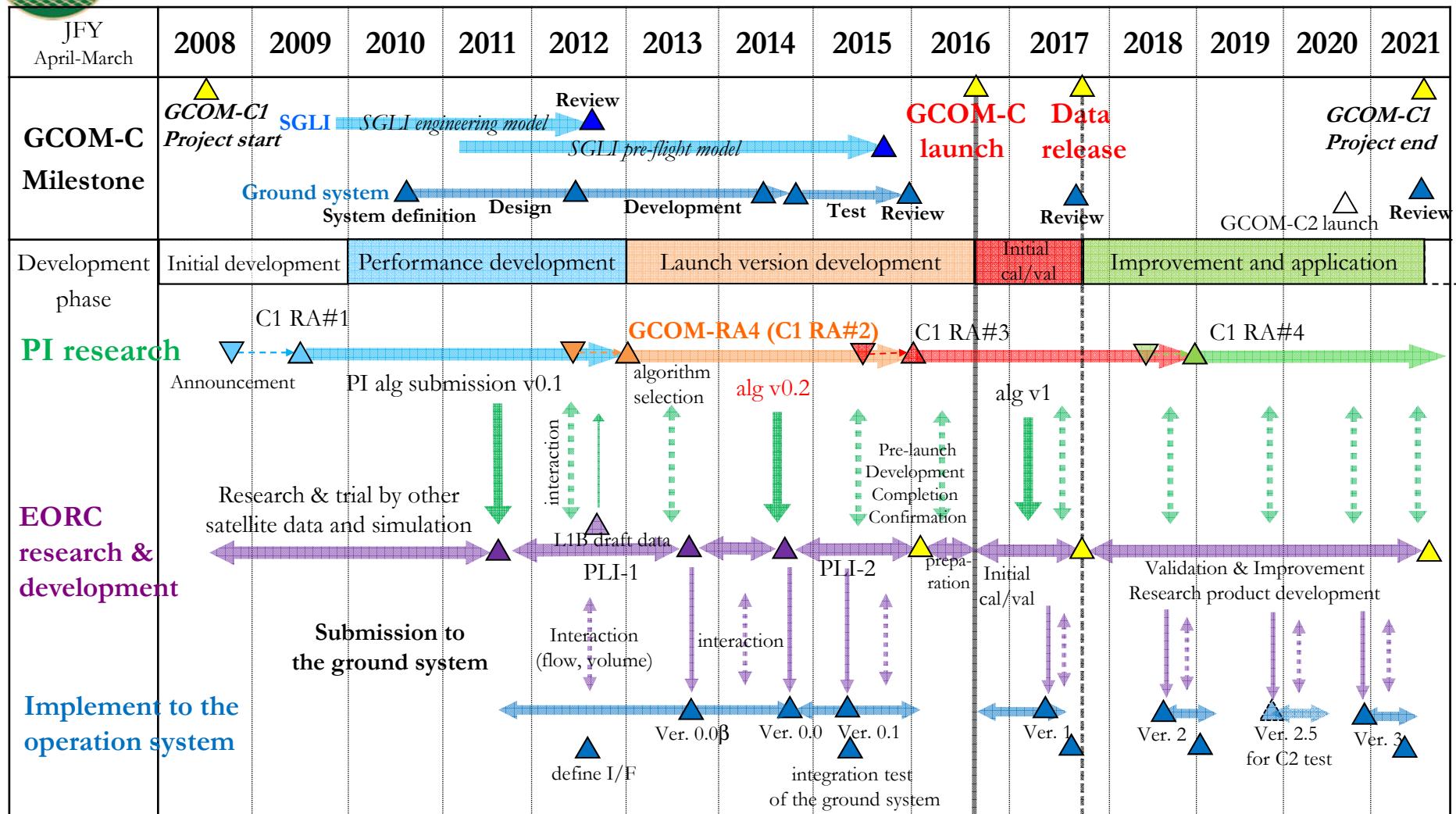
#### Standard and research products



- ✓ All standard products (L1, L2, L3) will be distributed by the internet (released to the public one year after the launch)
- ✓ Free for both science and commercial purposes



### 3. Product development: schedule





## 5. Summary

- The mission targets are contribution to the climate system researches, the carbon cycle and the radiative forcing, through series of satellites, GCOM-C, C2 and C3.
  - It has **250-m resolution** and **along-track slant-view polarization observation** to improve the land and coastal monitoring, and aerosol estimation.
  - On-orbit calibrations: diffuser (sun, lamp) with monthly moon observation
- GCOM-C/SGLI will be launched in **Japanese Fiscal Year 2016**.
- SGLI manufacturing and characterization, algorithms and ground processing system, and post-launch cal/val planning are being conducted by JAXA and GCOM-C PI team
- **The next RA will be announced in the summer 2015**
  - at-launch version algorithms, and post-launch improvement
  - vicarious calibration and product validation
  - application